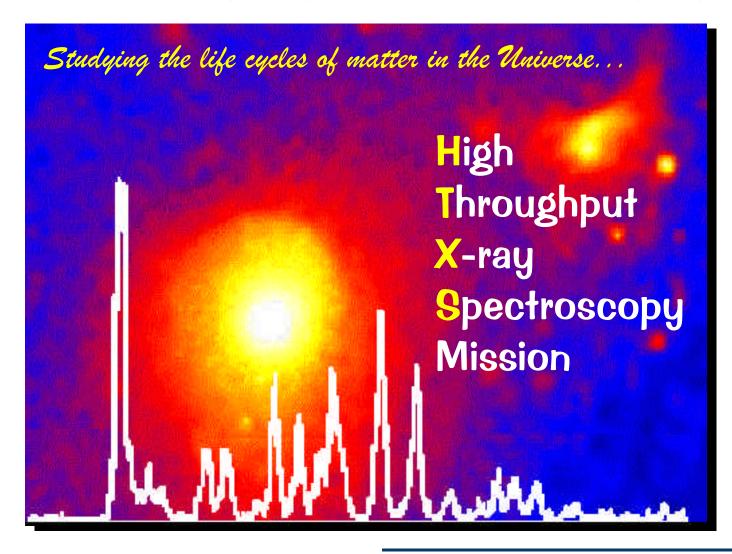


The High Throughput X-ray Spectroscopy (HTXS) Mission

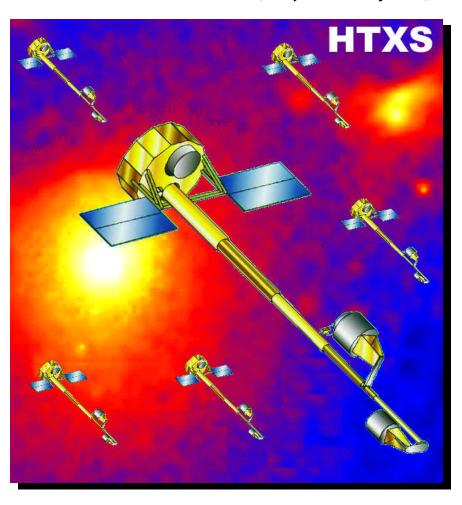
Nicholas White (GSFC) and Harvey Tananbaum (SAO)





High Throughput X-ray Spectroscopy Mission

Studying the life cycles of matter in the Universe



Key scientific goals

- Elemental abundances and enrichment processes throughout the Universe
- . Parameters of supermassive black holes
- Plasma diagnostics from stars to clusters

Mission parameters

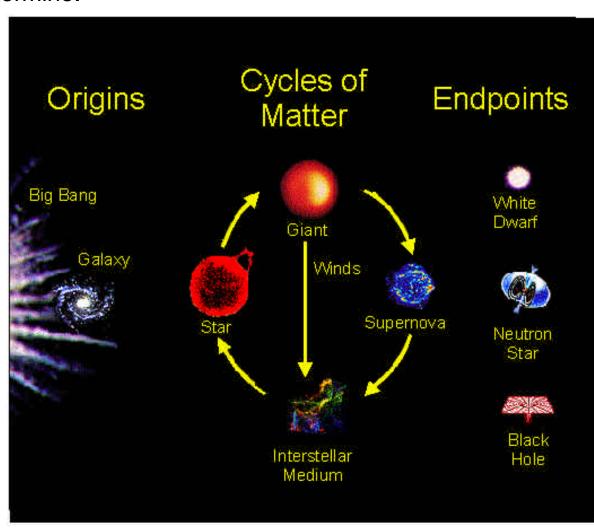
- Effective area: 15,000 cm² at 1 keV 150 times AXAF for high resolution spectroscopy
- Spectral resolving power: 3,000 at 6.4 keV
 5 times Astro-E calorimeter
- Band pass: 0.25 to 40 keV
 100 times increased sensitivity at 40 keV



Studying the Life Cycles of Matter with the HTXS Mission

Obtain high quality X-ray spectra for all classes of X-ray sources over a wide range of luminosity and distance to determine:

- the abundance of elements with atomic number between Carbon and Zinc (Z=6 to 30) using line to continuum ratios
- the ionization state, temperature, and density of the emission region using plasma diagnostics
- the underlying continuum process with a broad bandpass
- dynamics from line shifts and line broadening

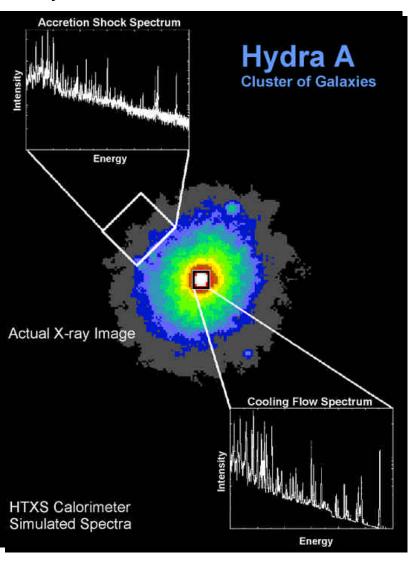






HTXS Observations of Clusters of Galaxies

Baryon content of Universe is dominated by hot X-ray emitting plasma

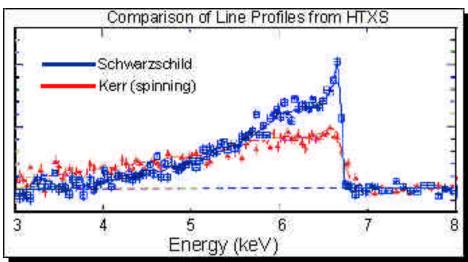


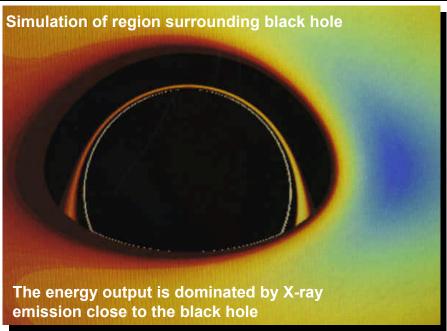
HTXS cluster observations essential for understanding structure, evolution, and mass content of the Universe

- Observe epoch of cluster formation and determine changes in luminosity, shape, and size vs redshift
- Measure abundances of elements from carbon to zinc, globally mapping generation and dissemination of seeds for earth-like planets and life itself
- Map velocity profiles, probing dynamics and measuring distributions of luminous and dark matter

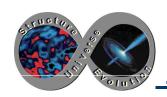


HTXS Will Determine the Nature of Super-Massive Black Holes

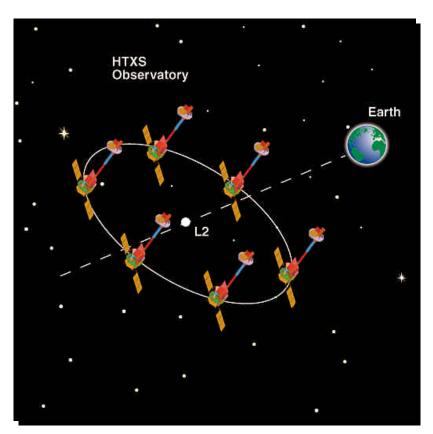




- HTXS will determine black hole mass and spin using iron K line
 - Spin from line profiles
 - Mass from time-linked intensity changes for line and continuum
- Trace black hole spin and mass over cosmic time from epoch of galaxy formation to present
- Relativistically broadened iron lines probe inner sanctum near black holes, testing GR in strong gravity limit



A Multi-Satellite Approach to Large Collecting Area



To achieve 15,000 cm² effective area on a single satellite requires a Titan-class launch.

An alternative low-risk approach to achieve large X-ray collecting area is to utilize six identical low-cost Delta-class satellites.

Launch intervals of three months.

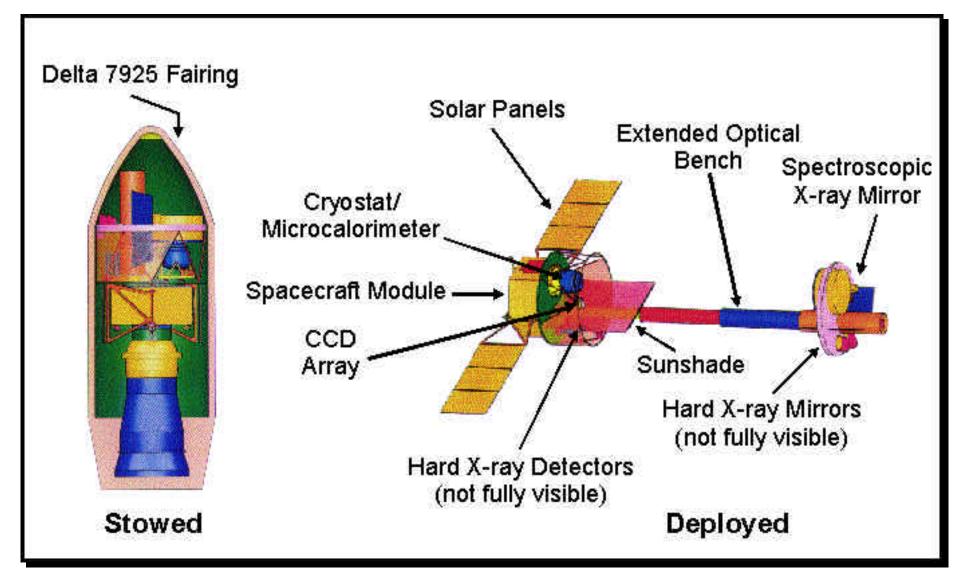
Facilitate simultaneous viewing and high efficiency by using libration point orbit.

The telescopes require a focal length of 8.5 m and use an extendible optical bench to allow a Delta-class launch.

Each spacecraft design lifetime is three years, with consumables targeted for a five-year mission.



HTXS Reference Design



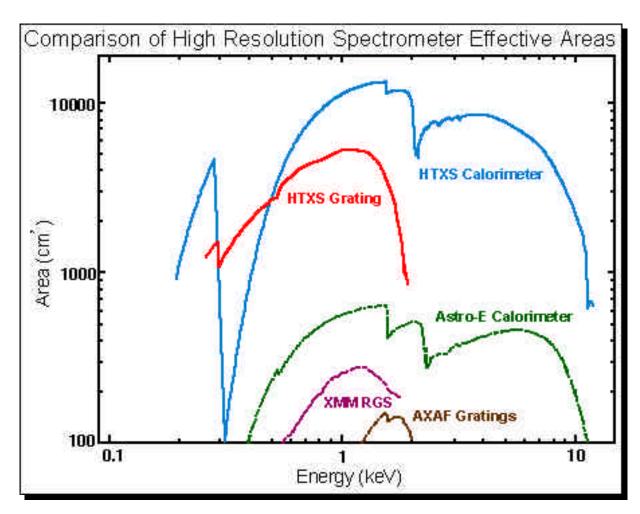


HTXS Requirements Flow Down

Science Goals Measurement Capabilities Key Technologies Minimum effective area: 15,000 cm² at 1 keV **High Throughput Optics** 6,000 cm² at 6.4 keV Elemental Abundances • Lightweight ó 250 kg 1,500 cm² at 40 keV and Enrichment throughout the Universe • Replicated shells and Telescope angular segments 15" HPD from 0.25 to 10 keV resolution: 1' HPD above 10 keV High Spectral Resolution • 2 eV microcalorimeter Parameters of arrays Supermassive Minimum spectral Coolers **Black Holes** resolving power ($E/\Delta E$): 300 from 0.25 to 6.0 keV • Lightweight gratings 3000 at 6 keV • CCD arrays extending to 10 at 40 keV 0.25 keV **Broad Bandpass** Plasma Diagnostics Band Pass: 0.25 to 40 keV • Multilayer optics from Stars to Clusters • CdZnTe detectors



HTXS Advanced Capabilities I. High Throughput



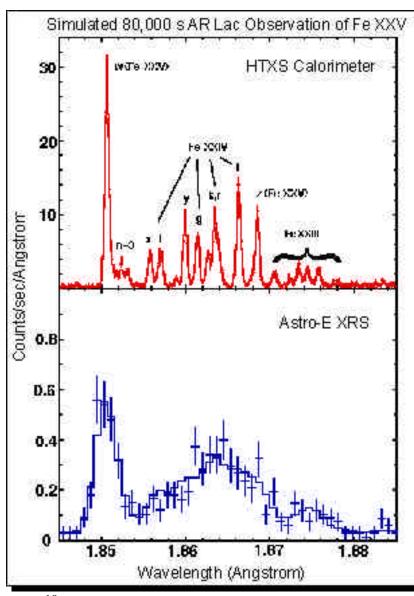
A 20-100 fold gain in effective area for high resolution X-ray spectroscopy

High throughput optics plus high quantum efficiency calorimeters

Lightweight reflection gratings maintain resolution and coverage at low energies (< 1 keV)



HTXS Advanced Capabilities II. High Spectral Resolution



The Next Generation Microcalorimeter Array

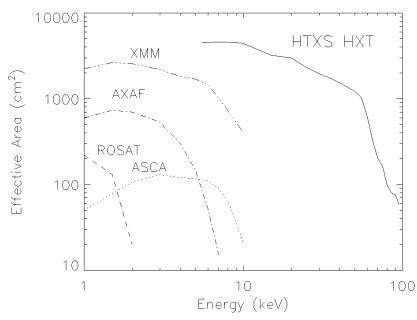
High quantum efficiency with the capability to map extended sources

- A factor of 5 improvement (to 2 eV) in spectral resolution
- Successor to the calorimeter to be flown on Astro-E (2000-2002)
- At Iron K, 2 eV resolution gives a velocity diagnostic of 10 km/s

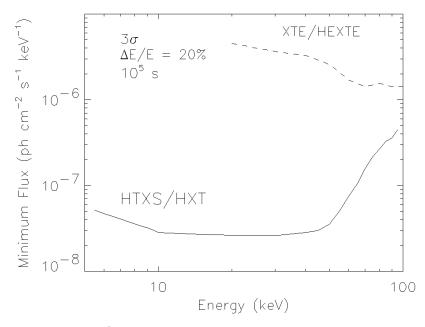


HTXS Advanced Capabilities III. Broad Bandpass

Multilayer coatings to enhance high energy response



Effective area as a function of energy for baseline HXT design.

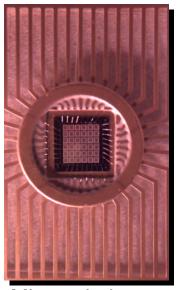


Continuum sensitivity as a function of energy for baseline HXT design

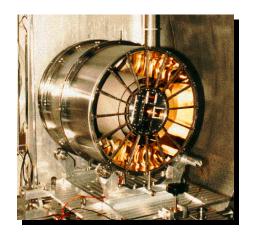
- No previous instrument has employed focusing in the Hard X-ray band
- Dramatic sensitivity improvements will be achieved



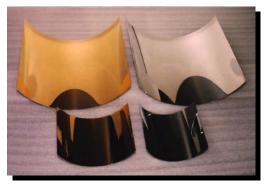
HTXS Technology Requirements



Microcalorimeters



Grazing Incidence X-ray Optics



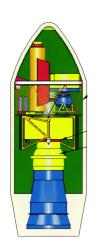
Multilayer Coatings



CdZnTe Arrays



page 12 Coolers



Deployable Structures



Autonomous Operations





HTXS Technology Roadmap & Mission Approach

- "Assembly line," low-cost concept for six identical satellites requires demonstrated performance for key technologies before phase C/D
 - essential for managing schedule and cost
 - requires investment ramping from ~ \$5M to ~ \$15M per year
- Technology roadmap developed by HTXS SWG, using expert teams and direct experience from AXAF, XMM, and Astro-E
- Coordinating with other cutting edge missions (FIRST, NGST, New Millennium, etc.)



Trade-offs: Angular Resolution vs Area

Imaging



0.1 arc sec 40,000 cm²

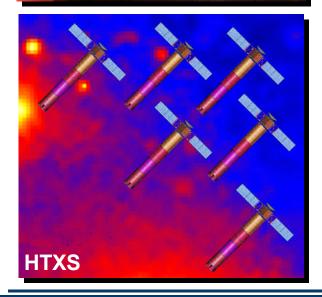


Spectroscopy

1 arc sec 780,000 cm²



0.6 arc sec 1,000 cm² (100 cm²)



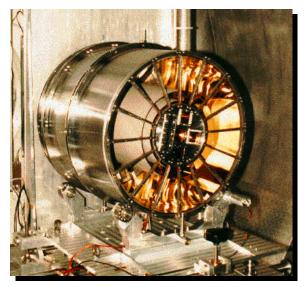
15 arc sec 30,000 cm² (15,000 cm²)

* effective area at the spectrometer





SXT X-ray Mirror Design Alternatives



Replicated Shells (e.g., XMM):

- meets 15" angular resolution
- requires factor of 10 weight reduction (2,500 kg --> 250 kg)
- investigate SiC, cyanate ester, and other lightweight carriers
- thin-walled rib-reinforced Ni shells

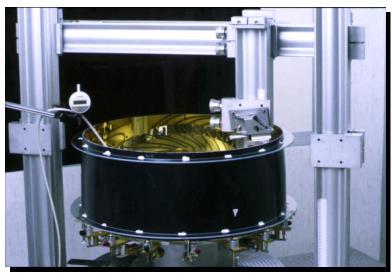


Segmented Optics (e.g., Astro-E):

- 210 kg weight meets the requirement
- requires factor of 4 improved angular resolution
- improved mandrels and foil alignment techniques



Spectroscopy X-ray Telescope Progress on Replicated Shells





Alumina (Al₂O₃) carrier shells produced in Italy at OAB -- lower temperature process, simpler, lower cost than SiC

- Carrier with 600 mm diameter and 3.2 mm walls produced by plasma spray
- Optical surface replicated successfully -- X-ray test in August 97
- Lighter weight alumina carrier -- same diameter,
 0.5 mm walls, three (3.3 mm) stiffening ribs
- Replicate this summer, then X-ray test

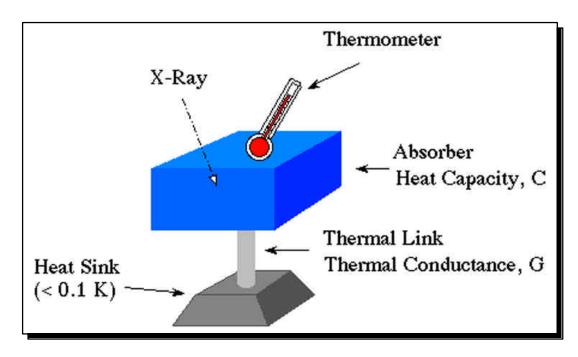
Progress on replicated shells at MSFC:

- Thin Ni shells with reinforcing ribs fabricated and ready for X-ray test
- Fabricated two 0.5 m diameter mandrels
- Modified equipment to handle 1.3 m diameter mandrels
- Awarded contracts for SiC and composite carriers



HTXS Technology Roadmap Microcalorimeters

Requirements on HTXS Microcalorimeter Array



A detector with 2 eV spectral resolution over the 0.3 - 12 keV band

- High quantum efficiency (~99% at FeK)
- Imaging capability commensurate with mirror PSF
 - 2.5' FOV => $30 \times 30 \text{ array}$
 - 10' FOV => 120 x 120 array
- Moderate speed for handling counting rates of 1 kHz or more

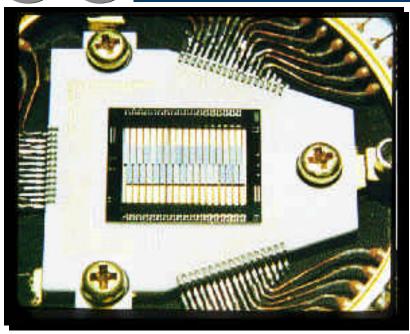
Current capability is 7-12 eV with 10 x 10 array

Technology developments required to achieve 2 eV resolution include

- more sensitive thermometers (transition edge superconductor)
- reduce heat capacity and power dissipation of existing system

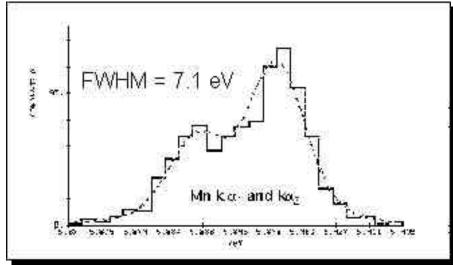


HTXS Calorimeter Advances



First flight test of Microcalorimeter

- Wisconsin/GSFC rocket flight 06/96
- 36 pixel array operating at 60 mK
- Observation of diffuse X-ray background
- Resolution of 14 eV at 277 eV achieved
- Detection of Sulfur IX and Oxygen VII
- Next flight 8/97 with improved array



First demonstration of TES Calorimeter at NIST

- Transition Edge Superconduction thermometer
- First result of 7.1 eV in Summer 1996 matches best to date
 - Capable of higher energy resolution Higher counting rates Lower cryogenic heat loads
- Not yet optimized!
 - expect significant improvement





HTXS Technology Roadmap Microcalorimeter Cooling System

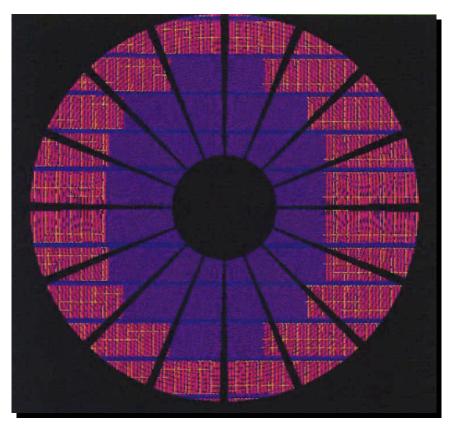
Develop long life, low weight, low cost, low vibration cooling systems



- Required Technologies
 - Mechanical cryocooler for thermal shields providing 10-100 mW cooling @ 3-5 K
 - Two-stage ADR system to reach 65 mK
- Investigate alternative technologies
 - Dilution refrigerator vs ADR
 - Sorption cooler vs Turbo-Brayton cooler
- Recent progress
 - Engineering model Turbo-Brayton 5 W, 65 K cooler run for 1.5 years with no degradation; being fabricated for 1999 HST servicing mission
 - 5-50 mW @ 4-10 K breadboard being fabricated with test in early 1998
- Require funding for two-stage ADR development



HTXS Technology Roadmap Grating/CCD Spectrometer



- The Grating/CCD spectrometer on HTXS will offer unprecedented sensitivity and resolution in the linerich, low energy (E < 1 keV) X-ray band.
- Effective area more than an order of magnitude better than that of the grating spectrometers on AXAF and XMM will be achieved.
- The design builds on the successful technical heritage of XMM and AXAF.
- Important new technology developments will include
 - Significant reduction in the mass per unit area of the grating array
 - Improved diffraction efficiency and reduced scattering from the individual grating elements
 - Significant reduction in the power consumption and total mass of the CCD and their associated read-out electronics
 - Improved low energy quantum efficiency in the CCDs





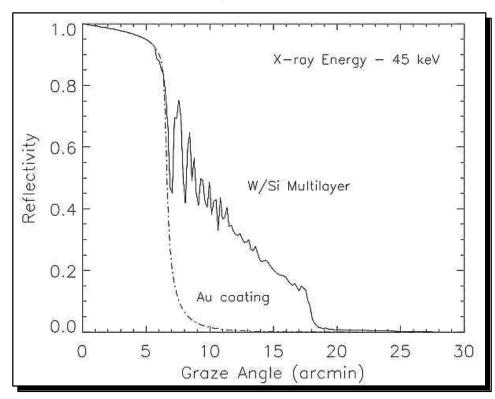
HTXS Technology Roadmap Hard X-ray Telescope: Optics

Primary Approach - Segmented shells

- Approach drawn from ASCA, ASTRO-E, SODART
- Epoxy replicated foils or thermally-formed glass substrates:
 - Mass ~ 100 kg achievable
 - Measured surface quality 3.7 Å glass, 5.5 Å foils meets requirements

Required technical development

Demonstrate coating without distortion

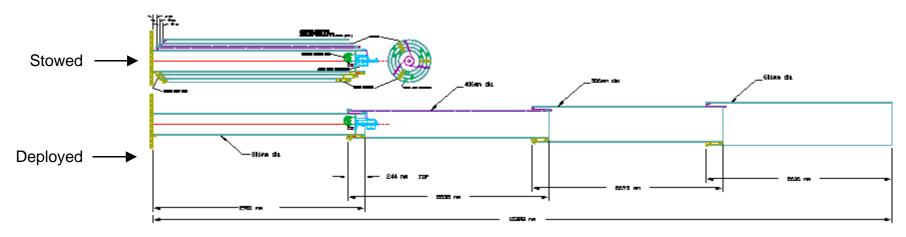


- W/Si multilayer on curved glass at Caltech/Columbia -- 200 layer pairs, 0.66 micron thick with acceptable stress
- Pt/C multilayer on an epoxy replicated foil mirror shell at GSFC/Nagoya -- 30 layer pairs, 0.13 micron thick with no distortion of foil due to stress
- Balloon flights planned in 1999
- Ray trace (left) indicates required area at 45 keV achievable



HTXS Extendible Optical Bench (EOB)

EOB provides the platform or support between the mirrors and the detectors and maintains their respective alignment for the mission



HTXS Requirements

- Provide for a focal length of approximately 8.5 meters with a stable structure (~ 1 mm) both mechanically and thermally
- EOB deployable to utilize Delta II-class launch vehicle
- Provide light tight protection to SXT and Grating/CCD

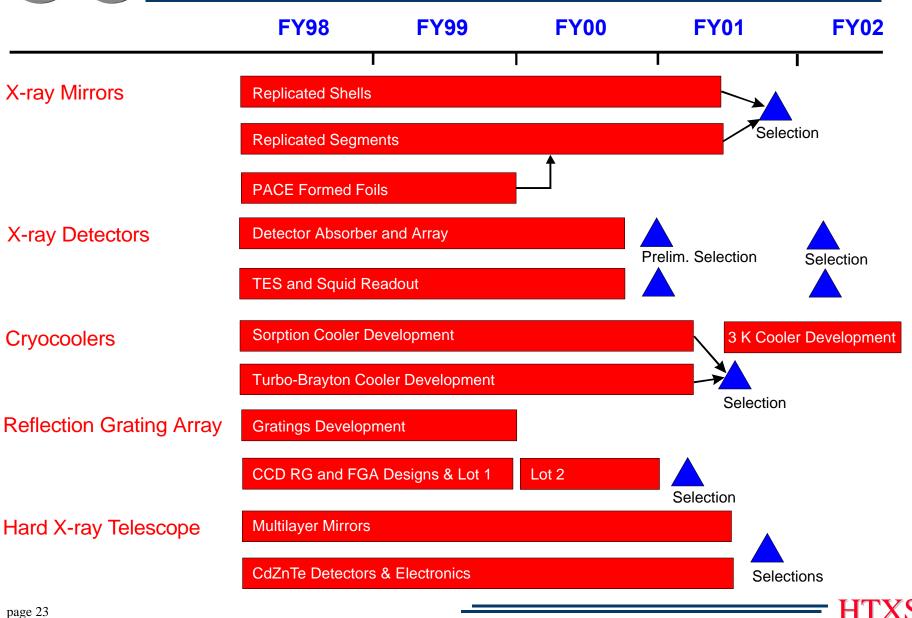
EOB Development Status

- Awarded partial funding through GSFC Directors Discretionary Fund in 97
- Optical alignment sensing system demonstrated in the lab using off-the-shelf components
- Vendors have reviewed baseline tube structures and confirm approach is feasible
- Next step is to produce an engineering unit to demonstrate the system performance





HTXS Technology Roadmap





HTXS Budget Requirements

Real Year \$ M

HTXS Mis s ion	1997	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008	2009	2010	2011	De ve lopment	Total
																Total	
Technology Develop.	0.6	5.6	11.1	14.3	15.4												47.0
Pre-Phase A	0.2	0.5	1.1	1.7	1.7												5.2
Phase A				0.7	2.4												3.1
Phase B						18.1	37.6										55.7
Phase C/D								78.1	108.2	111.3	114.4	73.5				485.4	485.4
Phase E											14.3	14.7	30.2	30.9	31.7		121.8
Launch Vehicle										65.4	67.2	69.0				201.6	201.6
Total	0.8	6.1	12.2	16.6	19.6	18.1	37.6	78.1	108.2	176.7	195.9	157.2	30.2	30.9	31.7	687.0	919.9
Above budget inflated	Above budget inflated to real year dollars																

- Additional cross cutting cryo-cooler \$1.6M in FY98
- Total development costs include Phase C/D and launch vehicle



HTXS Civil Service Manpower

Civil Service	1998	1999	2000	2001	2002	2003	2004	2005	2006	2007	2008
FTE	18	25	29	38	38	39	41	42	40	37	30

Notes and Assumptions:

- HTXS Civil Service (CS) Manpower estimate is a WAG; will be refined as mission definition, program structure, and acquisition strategy evolve
- This estimate assumes Phase C/D/E development performed by industry/universities
 - GSFC performs mission management/integration
 - CS manpower will grow if hardware elements (e.g., detectors) are brought "in-house" as a result of technology development



Mission Cost Fidelity

- Current C/D cost estimate for HTXS mission including all six spacecraft and launch vehicles is \$538K in FY97 dollars
- Cost estimate is "bottoms up" by major technology system
- Each system cost has been estimated based on other missions (AXAF, Astro-E, XMM) and/or industry input
- The HTXS cost estimates will be refined over the course of next year as the mission concept develops



The Outlook for HTXS

Summer of 1998...

- Technology development efforts have begun in earnest
- Mission concept study has demonstrated mission feasibility to next level of detail
- Cost estimates and Phase B/C/D schedule have been refined
- Acquisition strategy has been developed
- Outreach program underway

Summer of 2002...

- Phase B is halfway complete
 - Mission contractor has been selected
 - Systems Requirements Review has just taken place
- Technology developments required for HTXS are complete
 - Selections made between competing technologies



Associate Administrator Take-Away

HTXS traces the evolution of the Universe from origins to endpoints

Investment required to develop advanced technology to enable the mission

 assembly line production of lightweight, high performance optics, detectors, coolers, and spacecraft

Multi-satellite concept is low-cost, low-risk

Facilitates ongoing science-driven, technology-enabled extensions:

- spatial resolution,
- collecting area,
- o energy bandwidth, and
- spectral resolution